

Analytical Approach for Quantifying Noise from Advanced Operational Procedures

ASCENT Project 23

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Motivation

- Advanced operational procedures may have the potential to reduced aircraft noise and associated population exposure
 - RNAV/RNP, Continuous Descent Approaches, Delayed Deceleration Approaches, Climb Thrust Scheduling, etc.
- Traditional aircraft noise analysis assumes that engine noise dominates aerodynamic noise
 - Assumption may have been valid for earlier generation jet engines
- Current analytical approach does not fully capture noise impacts from aircraft configuration or other operational techniques which may have noise benefits
 - Integrated Noise Model (INM) / Aviation Environmental Design Tool (AEDT)

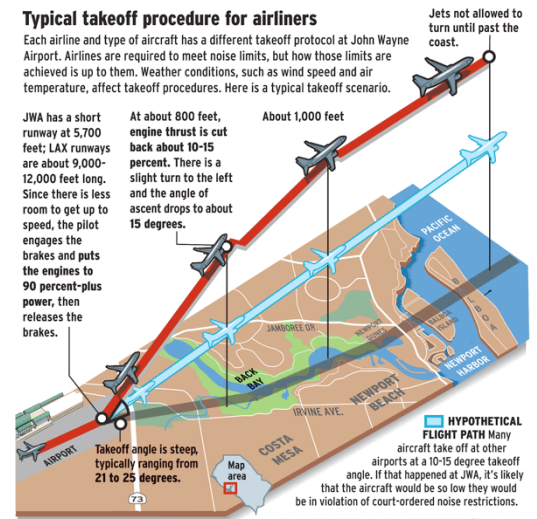
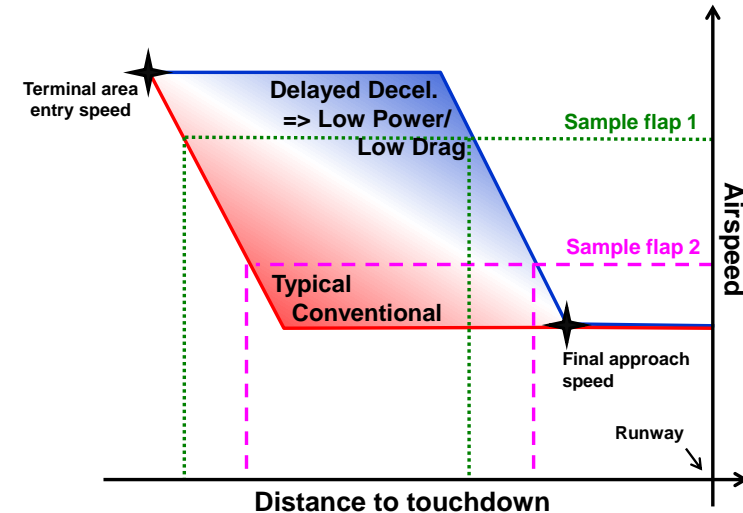
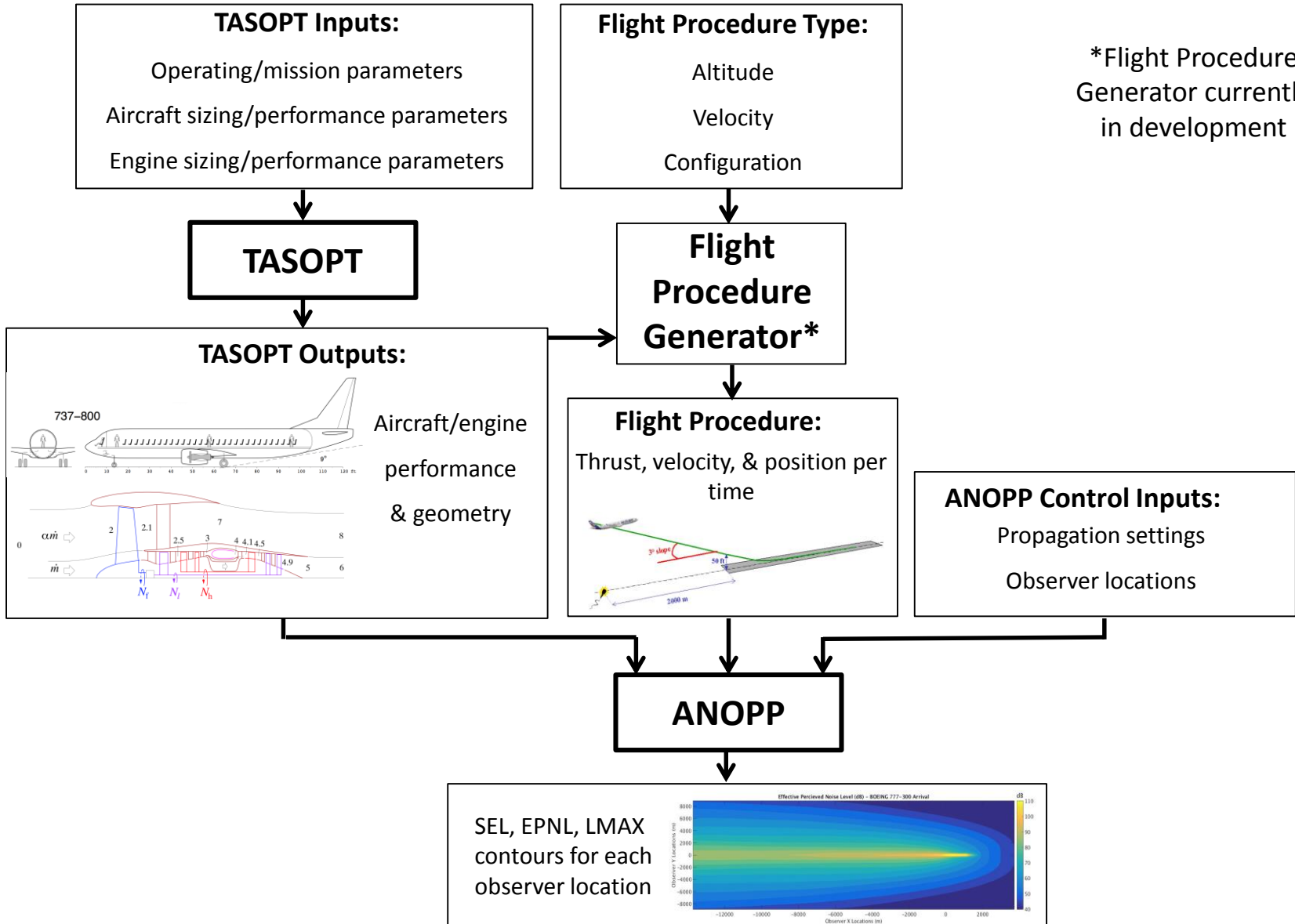


Figure: The Orange County Register

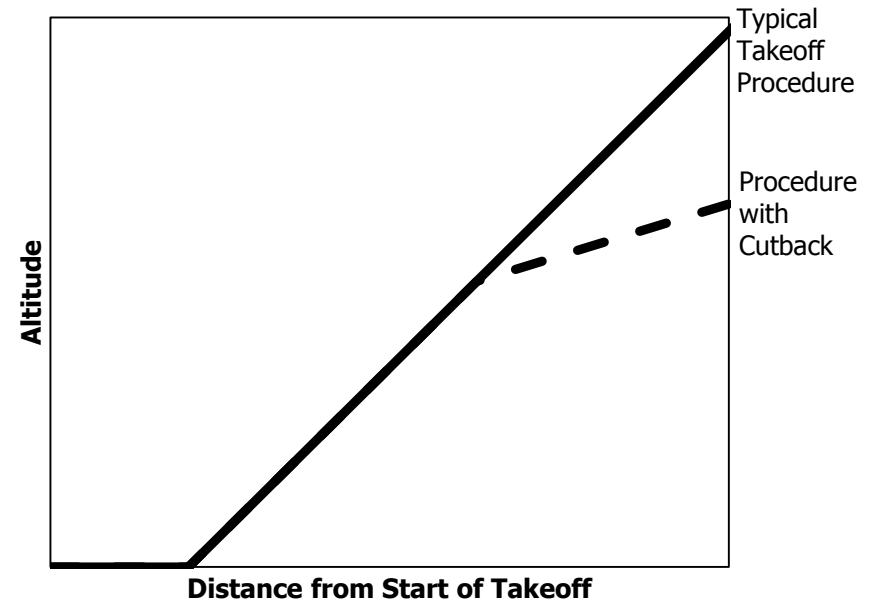
TASOPT - ANOPP Noise Analysis Framework



Example Application of Noise Analysis Tool: Thrust Cutback on Departure

- Typical takeoff procedure uses constant takeoff thrust throughout initial climb segment
 - Safety & efficiency benefits
- Thrust cutback after takeoff during initial climb can be used to reduce noise for nearby communities
 - Specific location of cutback determines overall impact of procedure

Variation of Departure Flight Profile with Thrust Cutback Location



Impact of Thrust Cutback Location on Single-Observer Departure Noise

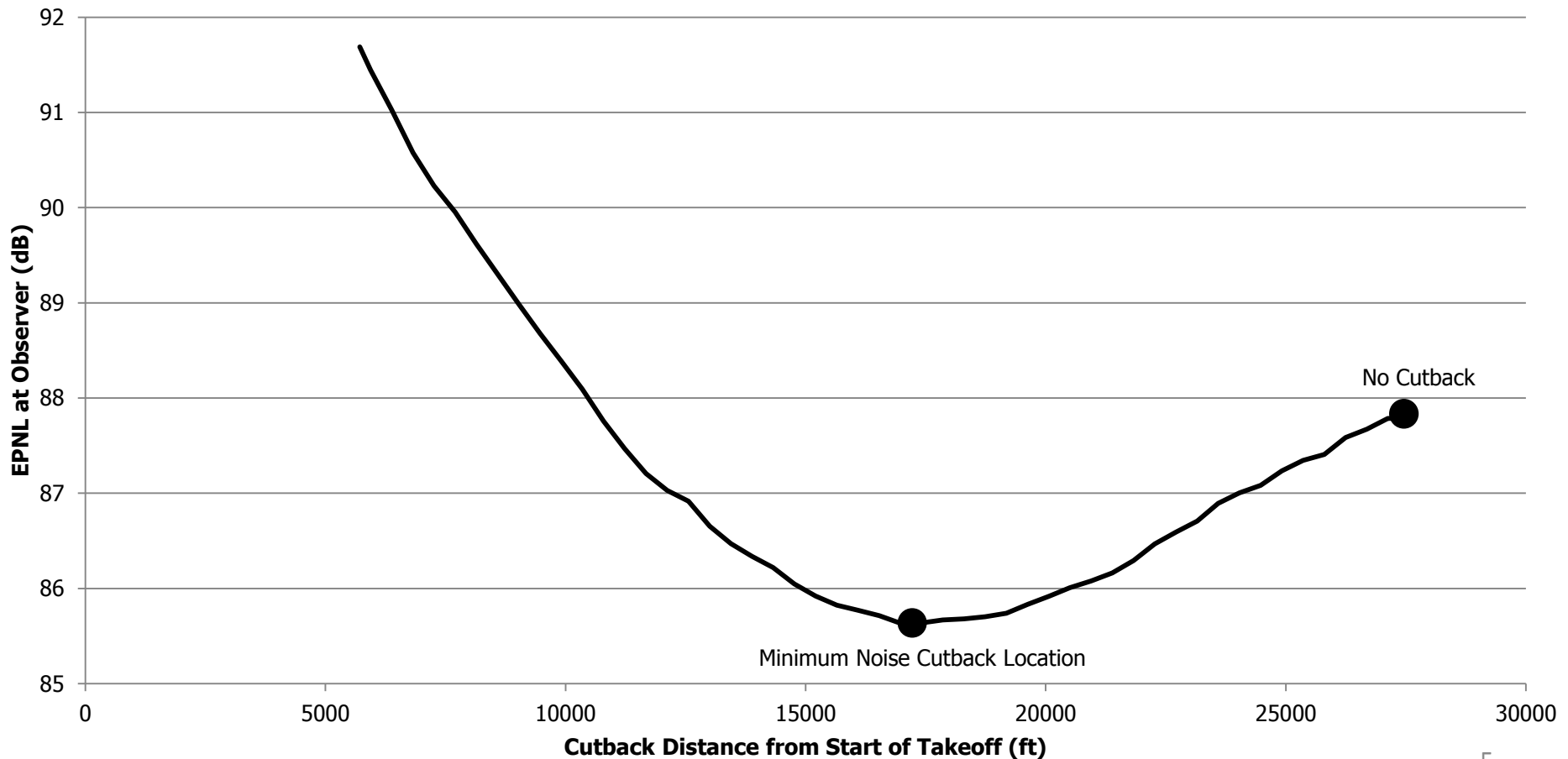


Boeing 737-800 Departures with Varying Thrust Cutback Location

Measurement Location: Extended Runway Centerline, 6.5km from Start of Takeoff Roll

Takeoff Weight: 172,300 lbs

Engine: CFM56-7B26



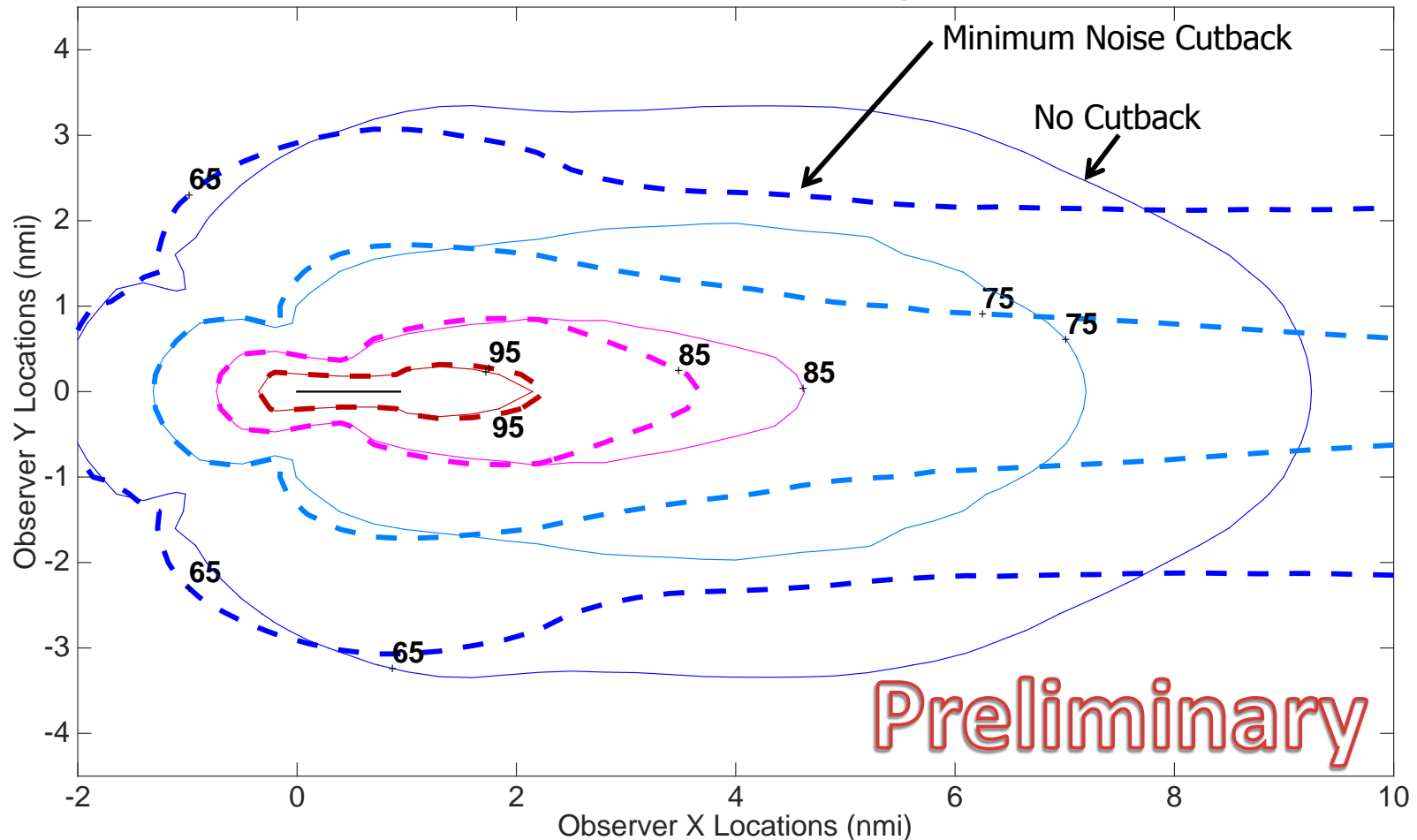
Impact of Thrust Cutback Location on Departure Noise Contour Geometry

Boeing 737-800 Departure Profiles

Takeoff Weight: 172,300 lbs

Engine: CFM56-7B26

Effective Percieved Noise Level (dB), Boeing 737-800 Departure

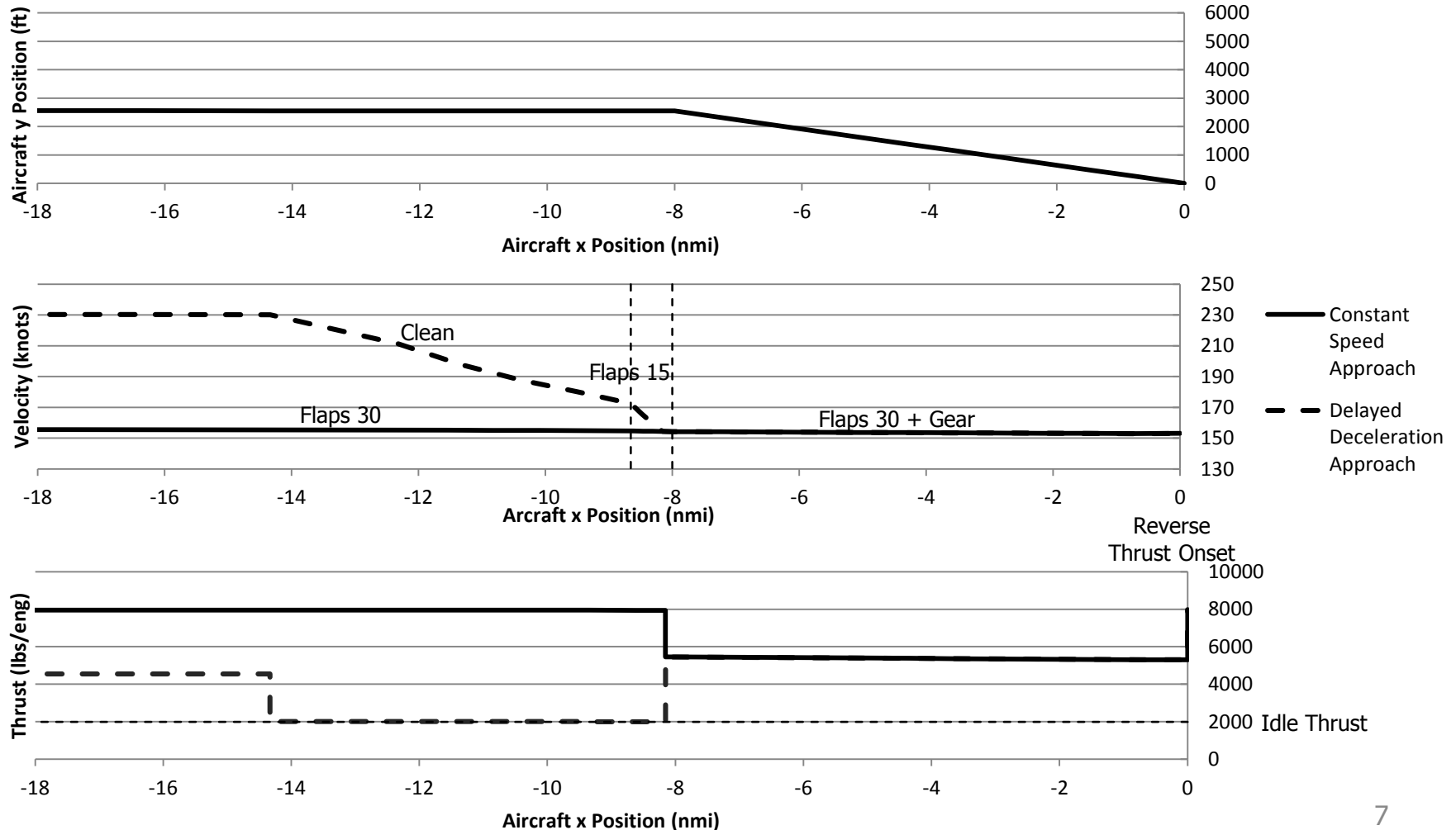


Delayed Deceleration Approach Profile: Glideslope Intercept from Level Flight

Boeing 737-800 Flight Profile

Landing Weight: 146,300 lbs

Engine: CFM56-7B26



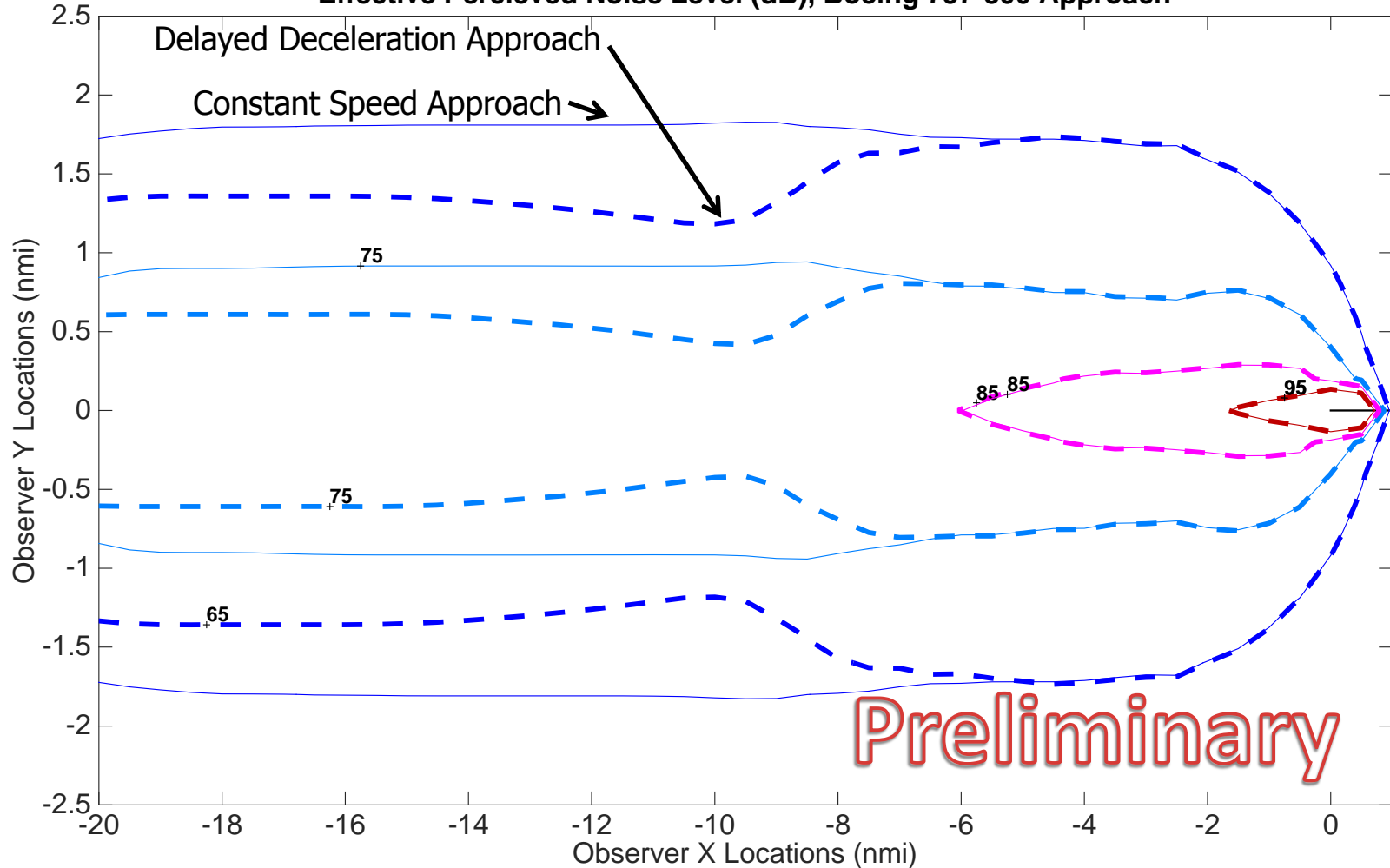
Impact of Delayed Deceleration Approach on Noise Contour Geometry

Boeing 737-800 Flight Profile

Landing Weight: 146,300 lbs

Engine: CFM56-7B26

Effective Percieved Noise Level (dB), Boeing 737-800 Approach



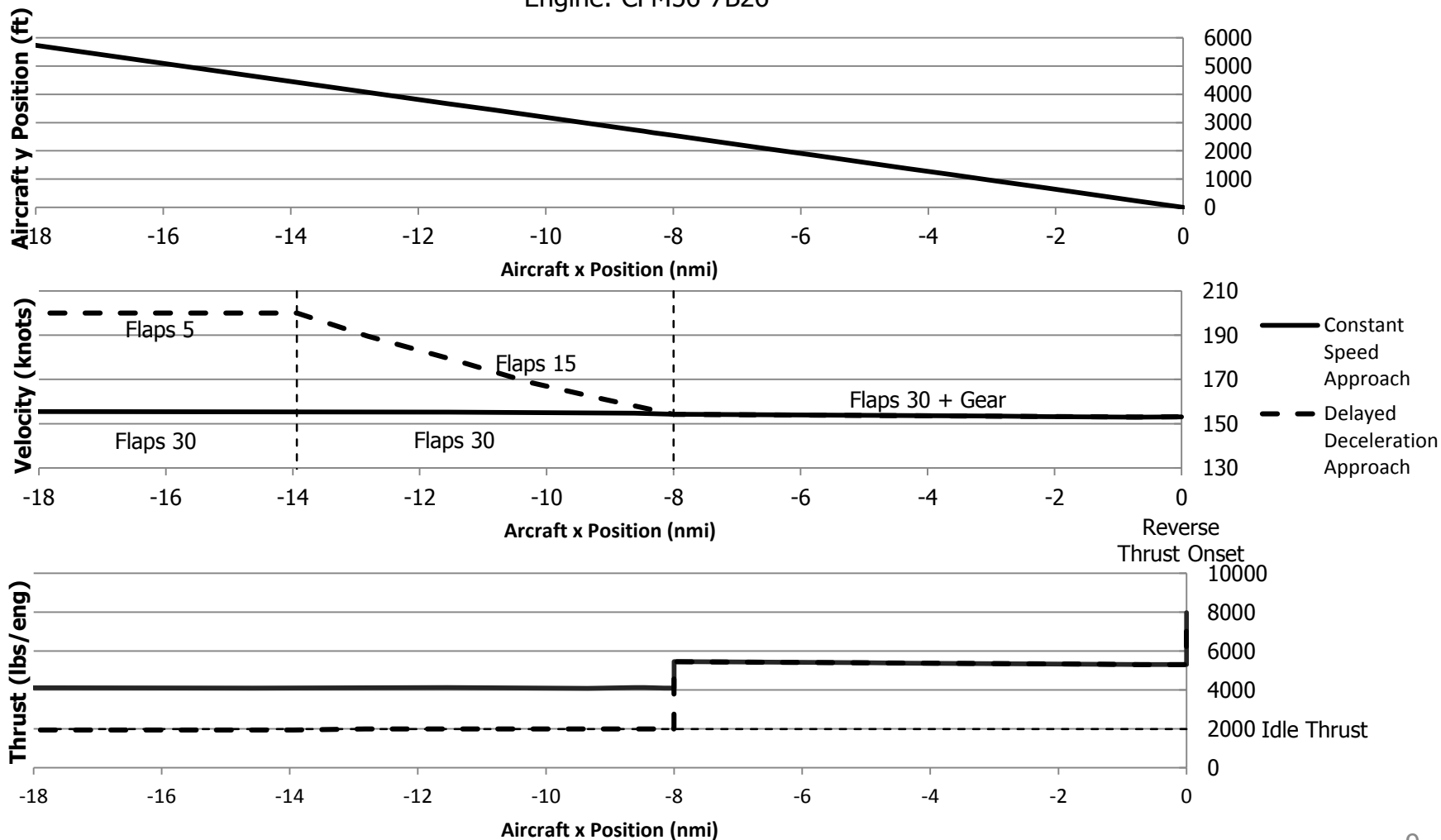
Delayed Deceleration Approach Profile: Continuous 3-degree Glideslope



Boeing 737-800 Flight Profile

Landing Weight: 146,300 lbs

Engine: CFM56-7B26



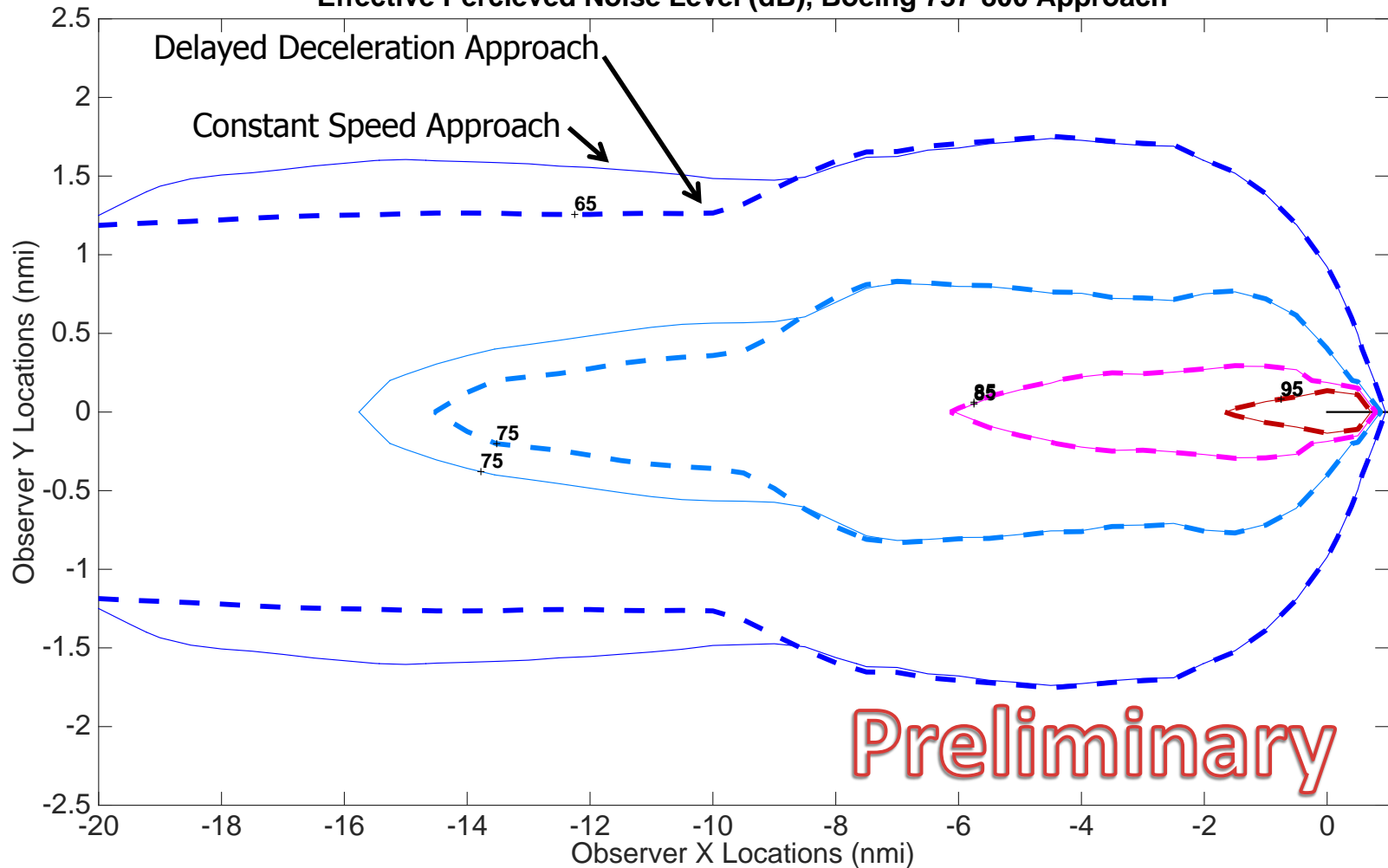
Impact of Delayed Deceleration on Noise Contour

Boeing 737-800 Flight Profile

Landing Weight: 146,300 lbs

Engine: CFM56-7B26

Effective Percieved Noise Level (dB), Boeing 737-800 Approach



Summary



- The objective of ASCENT 23 is to expand noise analysis capabilities such that benefits from advanced operational procedures (aerodynamic noise, masking, etc.) are reflected during the environmental review process
- Next steps:
 - Continue development of flight procedure generator
 - TASOPT refinements for improved drag modeling
 - Validation of aircraft noise models against noise certification test database
 - Implement representative fleet of aircraft types in full TASOPT/ANOPP noise architecture